	101	DUPLIC	CATE * For F	'AA 535 1	472		
Verified by	1 Operator #24		AVIATION AGE	68%)	Form at Budget	proved. Bureau No. 04–R060.	
	R REPAIR AND ALTERATIO				LER OR A	PPLIANCE)	
1. AIRCRAFT Grumman		G-21A		SERIAL NO.	1 /	N 88U	
				B-100			
NAME (First, middle, last) 2. OWNER Windjammer Air Taxi, Inc. B. COMPLETE ONLY FOR UNIT REPAIRED AND/OR ALTERED. DESCRIBE WORK ACCOMPLISHED ON REVERSE IN ACCORDANCE WITH CIVIL AERONAUTICS MANUAL 18.							
COMPLE CORDAN	TE ONLY FOR UNIT REPAIRE CE WITH CIVIL AERONAUTIC	ED AND/OR A S MANUAL 18	LTERED. DESCRI	BE WORK ACCOMPLIS	SHED ON I	OF WORK (Check)	
TINU	маке	MODEL SERIAL NO.		SERIAL NO.	MAJOR REPAI		
a. AIRFRAME	. AIRFRAME ************************************						
ь. POWERPLAN	т	The cistal airworthine		The detalateration identifies airworthiness requirements described ancient, subject	d herein come and is app as a confermity.	es with the applicable of crivings the applicable of the applicabl	
c. PROPELLER		authorized in CAR Part 18 section 12.11-5).					
d. APPLIANCE	TYPE AND MANUFACTURER	1/3/64 - C				FA Inspector	
4. AIRCRAF	T WEIGHT AND BALANCE DA ER the repairs and/or alterations described below were made.		spare component, in an aircraft. At th	ed by repair or alteration to will not be completed is time, it will be completed in the complete time.		However, in the component is in- installing agency,	
CATEGOR	Y EMPTY WEIGHT (Po			GRAVITY (Inches from datum)		SEFUL LOAD (Pounds)*	
Norma	al 6145 RMITY STATEMENT (Complete and	+ 22.1			1855		
	AME AND ADDRESS	. *	6. KIND OF AGENCY		c. CERTIF	ICATE NO.	
40 L	iam N. Kilborn ilac Lan e a, Florida - 33004	Foreign Cert Certificated Manufacture (Chec		on 1	A & P 1402309		
attach m	that the repair and/or alterations hereto have been made in a mation furnished herein is true	centannee with	i the reduirements of	I TAIL TO OF THE C. D. C.	d described Civil Air Re	on the reverse or gulations and that	
(Date r	2/64 epair and/or alteration completed)		- Mm	M. Holy (Signature to erutangia)	individual)		
Pursuant	VAL FOR RETURN TO SERVICE t to the authority specified below the Federal Aviation Agency ar	v the unit iden	tified in item 3 was	والمستوال المستشمل والمريدي			
3 0 8	PPROVED BY [FAA Designation FAA Fligh Standards Inspector		air Station To	anadian Department of ther (Specify) Inspec	tion A		
	ye of approval or rejection)		(Signature	of authorized individual; title	or recentments	1050	
	arded for engineering comment of the following the followi		tached memorandur		941 20	RECORD	
	GADO 5 (FAA designation number)		1 Mar	.(Signature Flight Standar	ds Inspector)	SO GADO	



This form must be completed in duplicate each time a major repair and/or alteration is made of an aircraft, airframe, power-plant, propeller or appliance. After the repair and/or alteration has been inspected and item 6 completed, the original copy of this form will be made available to the aircraft owner for retention as part of the aircraft records. The duplicate copy is retained by the FAA for administrative purposes.

See CAM 18 for detailed instructions concerning the information to be furnished with this form and instructions concerning its preparation.

8. DESCRIPTION OF WORK ACCOMPLISHED.

Removed all existing redio equipment, modified instrument panel, and installed redios as follows:

Two ARC T-11B transmitters install ed forward of pilots rudder pedals at arm -46. Transmitter control box mounted under pilot's instru ment panel at arm -21. Transmitting anten na is located just ahead of windshield. Model 888 VOR indicator and Kollsman model 1859 course selector installed in panel at arm -27. ARC 022A control unit installed in panel at arm -26. ARC R-13B reciever and B-10A convertor installed at arm -62, and E-13B rack installed at arm -61 forward of bulkhead 7 on right side. ARC F-10A filter amplifier installed at arm -61. Exfi sting recieving antenna was used.

Removed all existing seats, replaced floorboards, and installed seats as follows:

Installed 5 seats in cabin at locations shown on attached loading charts. Seats attached to structure same as Grumman original. 3 pl ace lou nge (STC SA2-760) installed on right side of cabin to original Grumman seat attach points, location shown on attached loading charts. Seat bel ts for lou nge attached to original Grumman seat attach points with 1/8" aircraft cable and AN 43 eye bolts.

All work done according to the ap plicable parts of CAM-18.

Weight and balance report and loading charts are attached.

An electrical load analysis was accomplished in accordance with CAM-18.30-12(i). The ammeter reading with all continous loads on was 15 amps. Adding 10% this comes to 16.5 amps. This is within 80% of the cap acity of bo th generators or one alone since this airplane is equipped with two 50 amp generators.

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*If additional space is needed attach additional sheets bearing alreast nationality and registration mark and date work completed.

Check block if additional sheets are attached.

*U.S. GOVERNMENT PRINTING OFFICE: 1961 O -587360

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Form FAA-837 (4-52)